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GUJARAT TECHNOLOGICAL UNIVERSITY

B.E. Sem-Vth Examination December 2010

Subject code: 150104

Date: 18 /12 /2010

Sub Name: Computational Fluid Dynamics-I

Time: 03.00 pm - 05.30 pm

Total Marks: 70

Instructions:

- 1. Attempt all questions.
- 2. Make suitable assumptions wherever necessary.
- 3. Figures to the right indicate full marks.
- Q.1 (a) Derive the energy equation for a viscous flow with heat transfer in non 07 conservation form.
 - **(b)** Justify CFD is a research tool. Write and explain the steps involved in CFD **07** process.
- Q.2 (a) What is the difference between the Euler's model and Navier stokes model of 07 equations? Write the generic form of Navier Stokes model.
 - **(b)** Explain the supersonic flow over the nose of a blunt body. Write a short note on **07** shock capturing method.

OR

- **(b)** Consider the irrotational, 2-D, inviscid, steady flow of a compressible gas. The flow field is slightly perturbed from free seam like flow over a thin profile. Find the roots of equations involved in such kind of flow problem using Cramer's rule and Eigen method.
- Q.3 (a) Distinguish between the basic discretization techniques. Derive the expression for 1st order forward, 1st order rearward and 2nd order central difference equation with respect to x.
 - (b) Explain the domain and boundaries for the solution of parabolic equation in 2-D. 07 also explain the steady boundary layer flows over a body.

OR

- Q.3 (a) Derive the differential equation to unsteady 1-D heat conduction equation. Also 07 define the accurate solution and precise solution.
 - **(b)** Write a short note on Lax Wendroff method and gives its stability criteria.
- Q.4 (a) Differentiating between explicit approach and implicit approach for the solution of difference equations. Formulate the explicit form for 1-D heat conduction equation.
 - **(b)** With an example explain the concept of compressed grid.

OR

- Q.4 (a) Describe Mac Cormark multi-step method.
 - Consider the viscous flow of air over a flat plate. Variation in velocity with respect to y is given as: $u=1582(1-e^{-y/L})$. Where L=1 unit and $\mu=3.37\times10^{-7}$ slug/ (ft.s). y is from 0 to 0.3 in the steps of 0.1. Find the percentage error in shear stress, involved in 1st ordered and 2nd ordered difference compared to exact solution.
- Q.5 (a) Draw and explain the subsonic-supersonic flow through the C-D nozzle and also 07 show the variation in properties along the length of nozzle
 - (b) Derive the momentum equation for the 2-D subsonic supersonic flow through C-D 07 nozzle.

OR

- Q.5 (a) How the boundary conditions and initial conditions are applied to the nozzle flow. 07
 - (b) Why the governing equations are to be transformed into non dimensional form? 07 Derive the momentum equation in non dimensional form for the nozzle flow.
